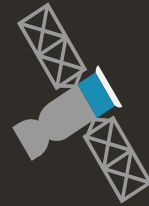
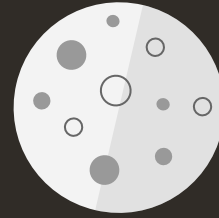


# Beyond Fuel

An On-Orbit Refuelling and Debris Mitigation Solution

28<sup>th</sup> May 2024





# Our Journey



**01**

**The Growing LEO  
Ecosystem**

**02**

**The Limitation of  
Satellite Lifetimes**

**03**

**The Beyond Fuel  
Solution**

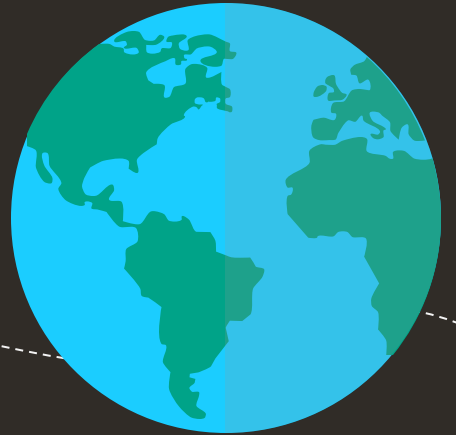
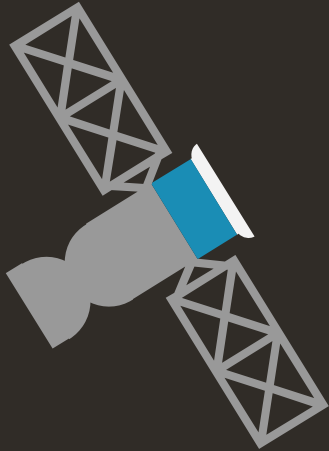
**04**

**Project Overview  
and Next Steps**

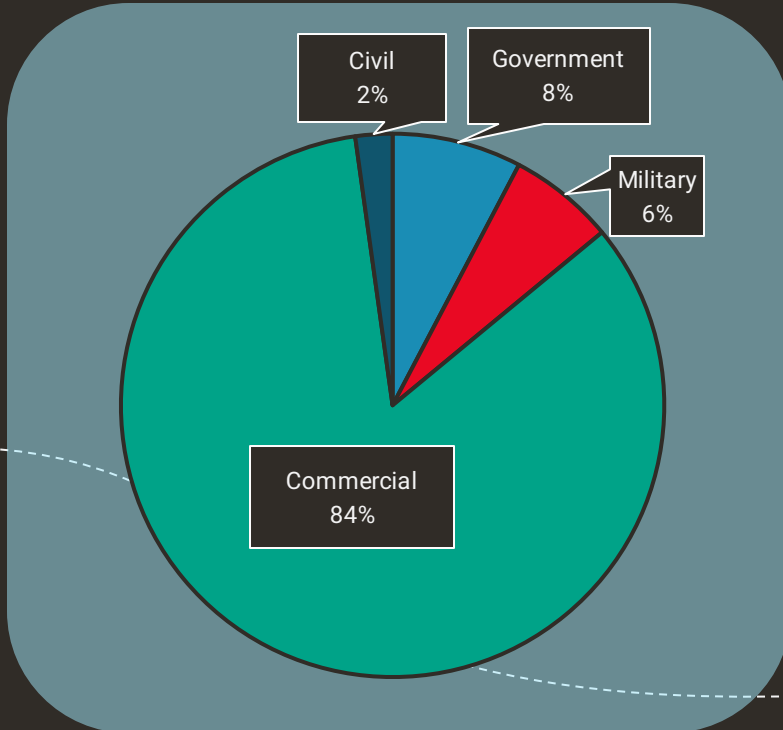


01

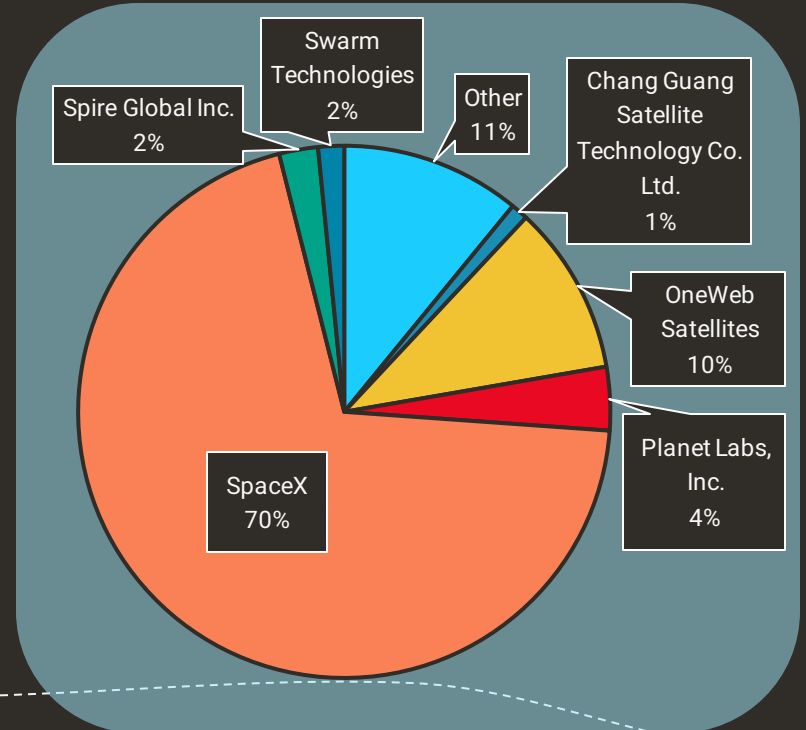
# The Growing LEO Ecosystem



# Satellite Surge – The Growing Market of Low Earth Orbit

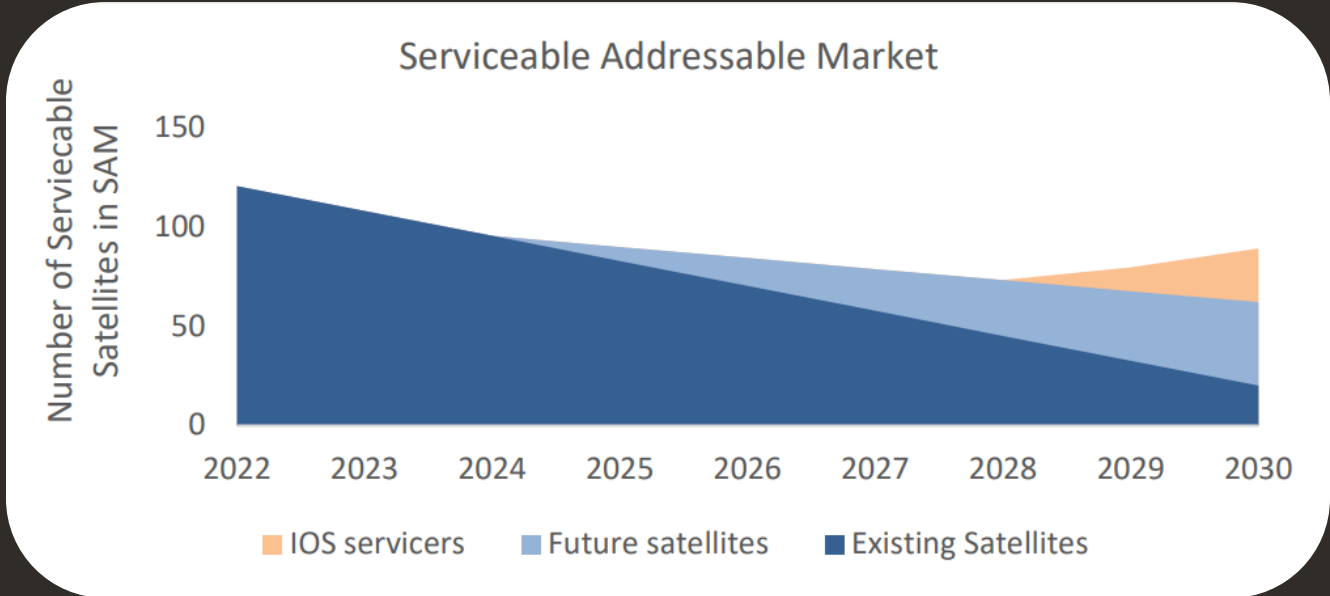


Satellite Sectors By Population



Commercial Sector By Company.

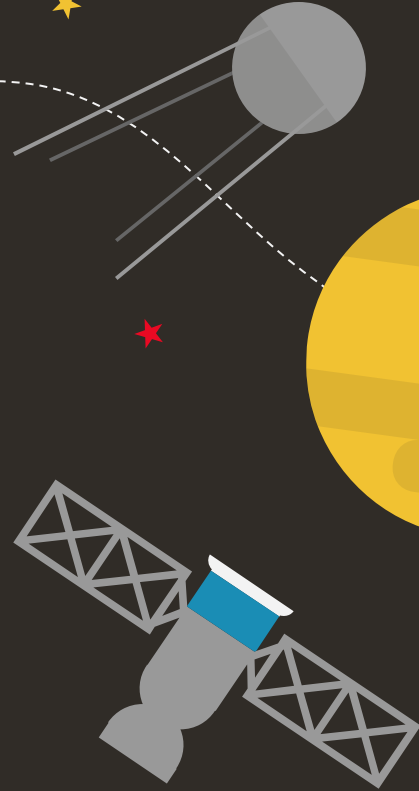
# Refueling Satellites? In this Economy?



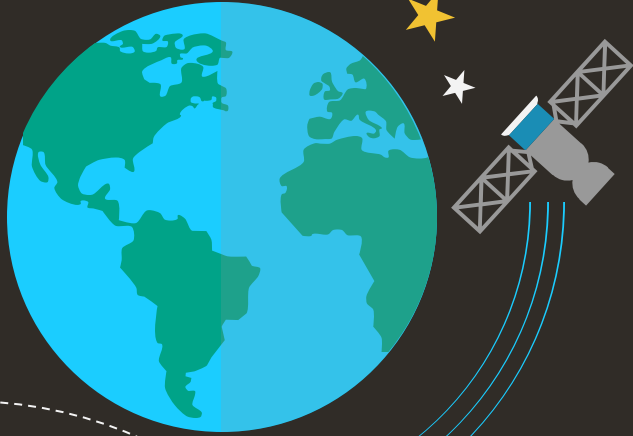
Service addressable market for existing and future satellites

02

## The Limitation of Satellite Lifetimes



# The Lifetime Problem

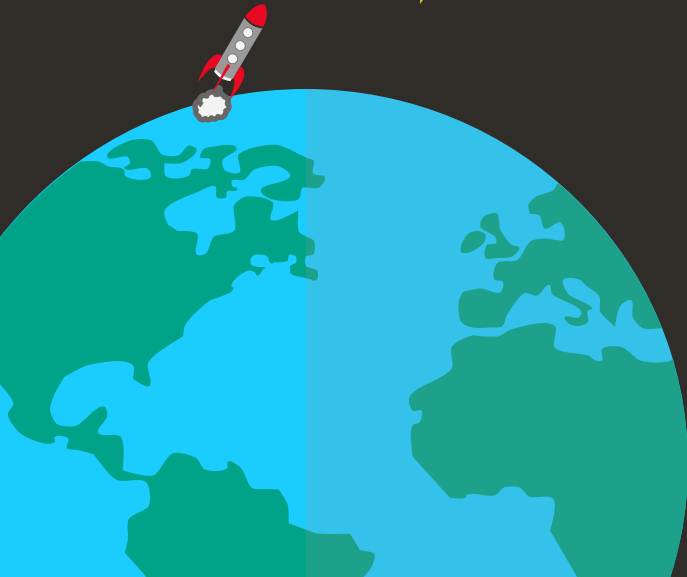


- **82%** of LEO satellites have lifetimes below **4 years**.
- **Limited fuel reserves** are utilised to conduct station keeping.
- **Fuel depletion or debris impact** lead to decommission.

# Replacement, Yes – But At What Cost?

- **\$400k** per satellite per launch.
- **\$500k >** per satellite per unit.
- **~\$1 Million** per satellite replacement.

Assuming the satellite to be replaced is a 240kg Starlink satellite ridesharing on a Falcon Heavy and estimating manufacturing costs. Values from 240kg Starlink Satellite







# A Better Future; On-Orbit Servicing

- NASA, Robotic refuelling missions.
- ESA, 'Clean Space' initiative.
- Astroscale, ELSA-d.

The background is dark grey with several decorative elements: a red star in the top left, a yellow star in the top center, and a yellow star in the top right. A white dashed line curves across the top. In the bottom left, there is a red star, a yellow star, and a white star. A white dashed line curves across the bottom. On the right side, there is a grey moon with various sized circles representing craters, a yellow star, a red star, and a white star. A white dashed line loops around the moon.

**03**

# The Beyond Fuel Solution



# Project Work Streams

1

Orbit Determination,  
Maintenance and  
Rendezvous

2

Robotic Arm Control  
and Material Selection

3

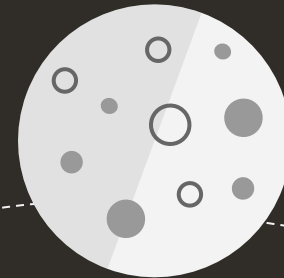
Fuel Pumps in  
Microgravity

4

Electrodynamic Tethers  
for Debris Mitigation

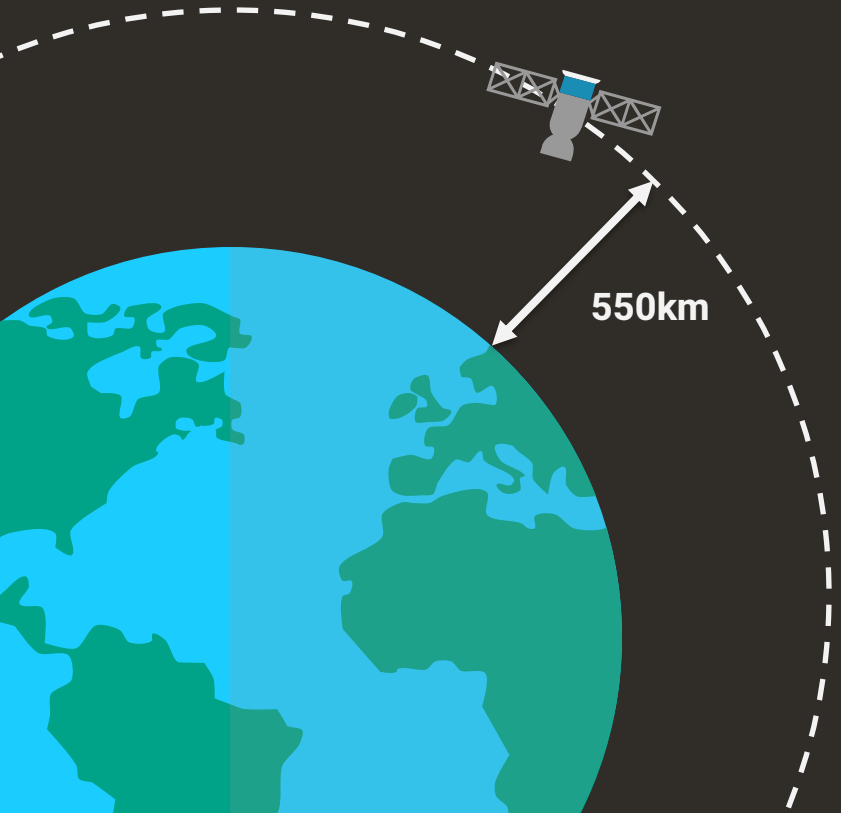
5

Structural Integrity  
and Material Selection





# Determining An LEO Parking Space

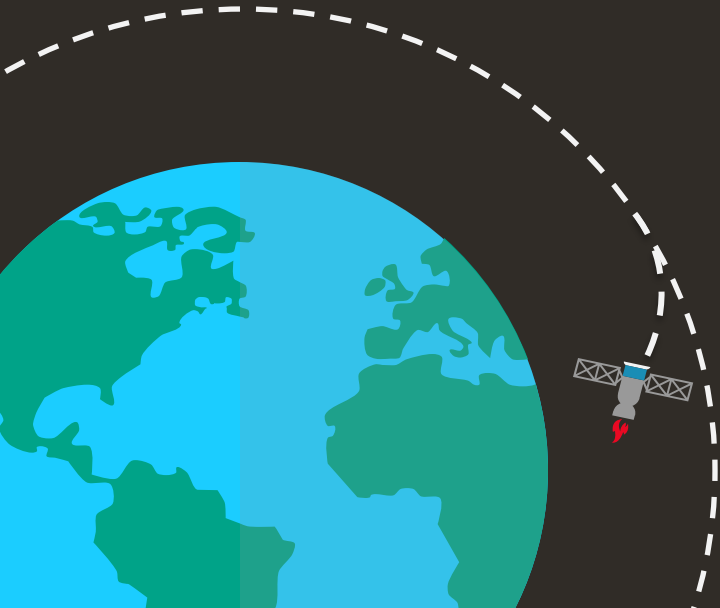


- **Low Earth Orbit**
- **550 km from surface**
- **53° inclination**
- **0 eccentricity**



# The Cost Of Longevity

- Across a **20-year period, 104 kg of fuel is required** for station keeping.

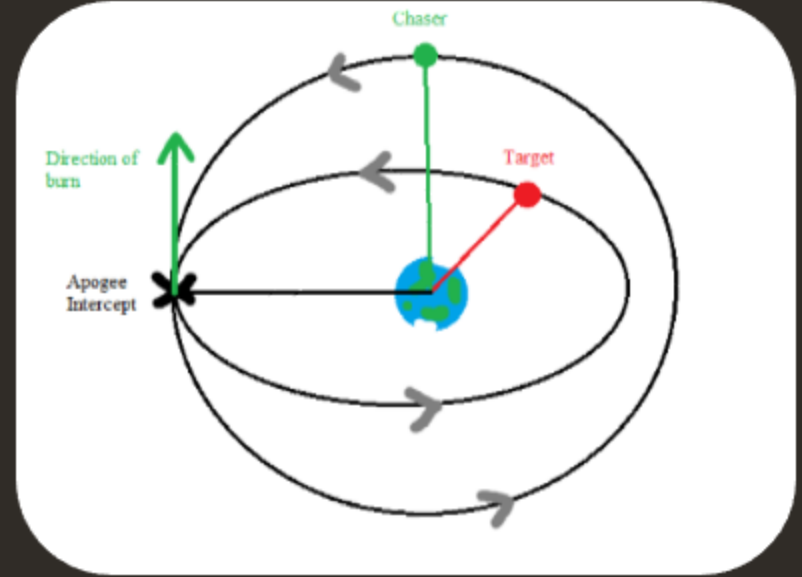
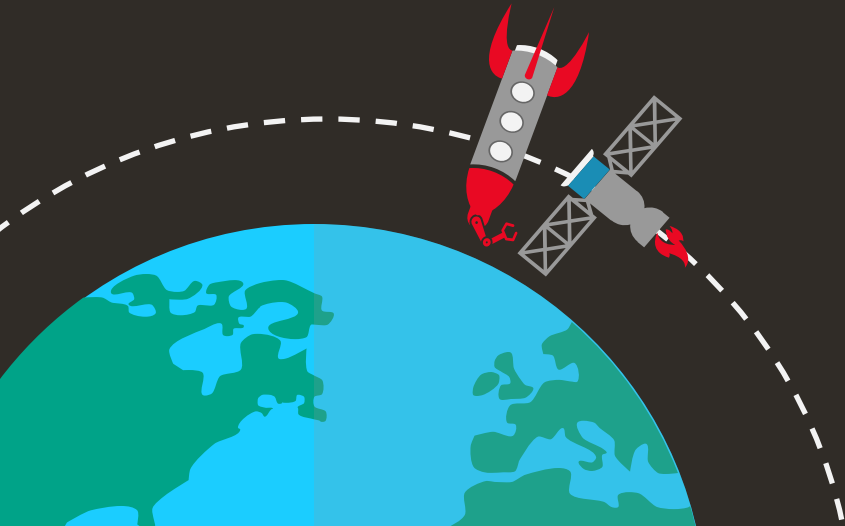


Visualization of an orbit maintenance **Hohmann transfer**.



# High-Cost Hookups: Rendezvous

- For an exemplary rendezvous mission, 182 kg of fuel is required.



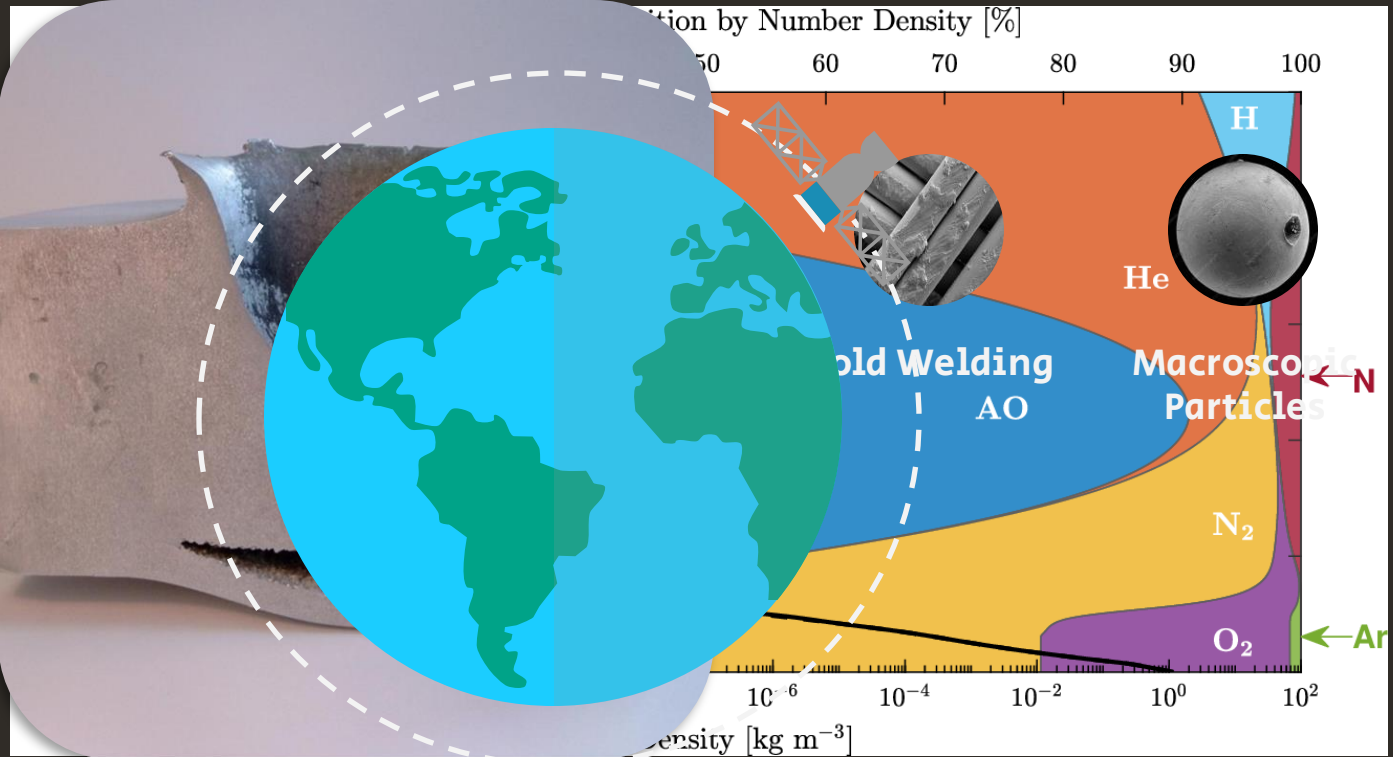
Plan for a **Hohmann Spiral Transfer** for long range rendezvous.



# The LEO Environment



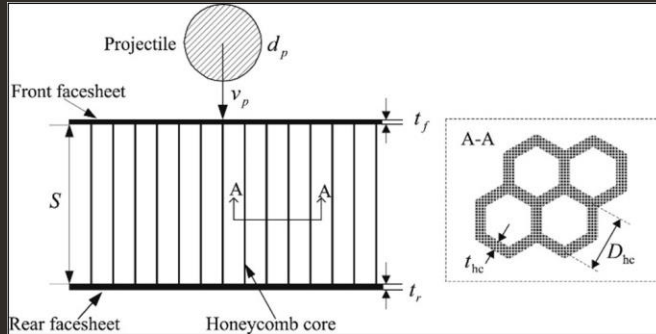
Atomic Oxygen



# Robot Arm Materials



## Outer Protective Material

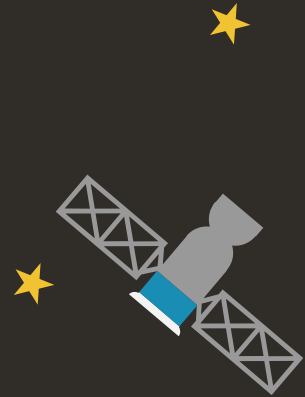


**Nextel + Aramid Fibres**

## Inner Structural Material

- Specific Strength
- Fatigue Resistance
- Thermal Expansion
- Specific Stiffness
- Specific Price

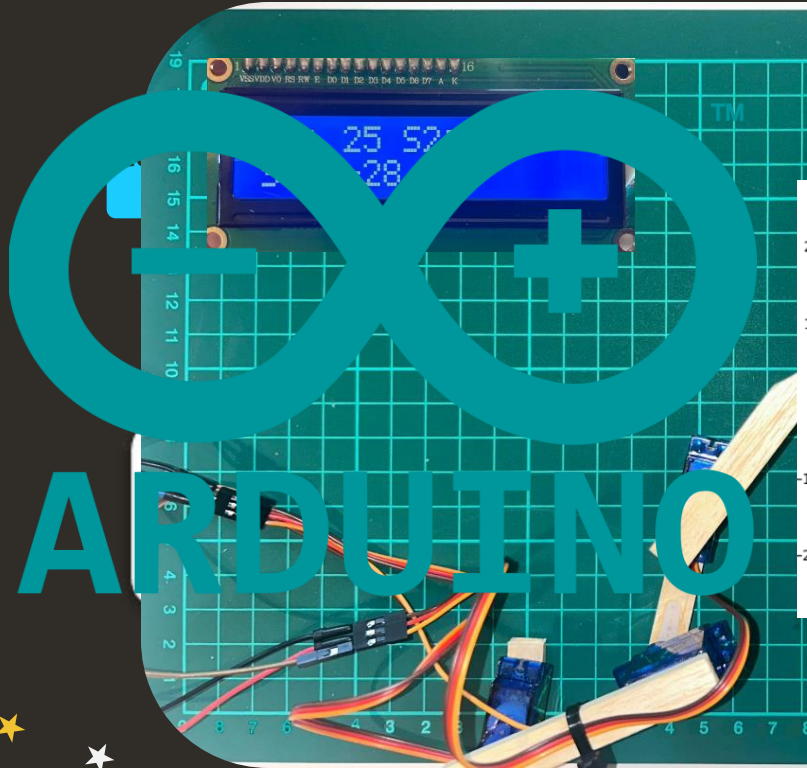
**PEEK IM Carbon Fibre**







# Robot Arm Control

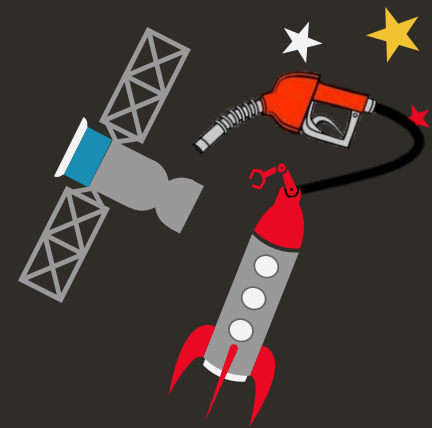


Solution 1:  
 S1 Angle: 24.52 degrees  
 S2 Angle: 55.00 degrees  
 S3 Angle: -28.18 degrees

Solution 2:  
 S1 Angle: 78.16 degrees  
 S2 Angle: -55.00 degrees  
 S3 Angle: 28.18 degrees



# Fuel Pumps in Microgravity

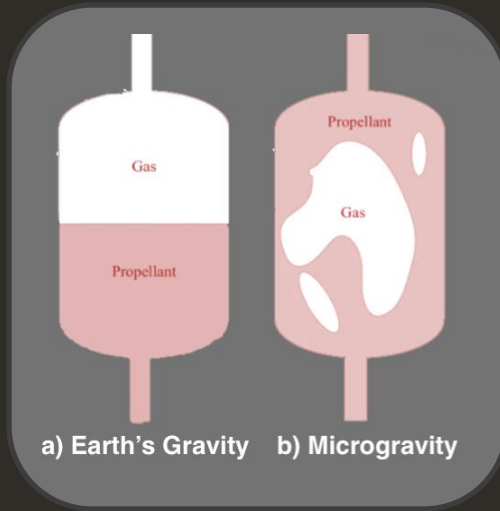


## Challenges

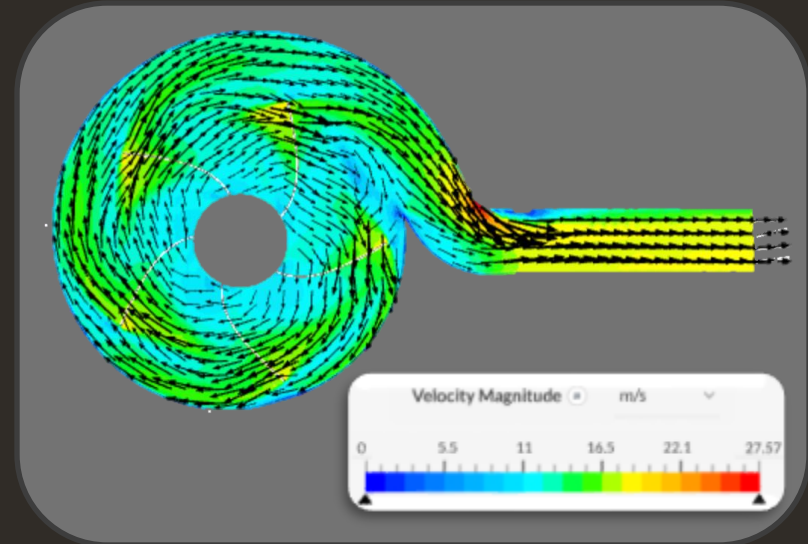
- Microgravity
- Surface tension
- Bubbles forming due to sloshing of fuel in the tank

## Solution

- Centrifugal pump



Surface Tension at Different Gravities

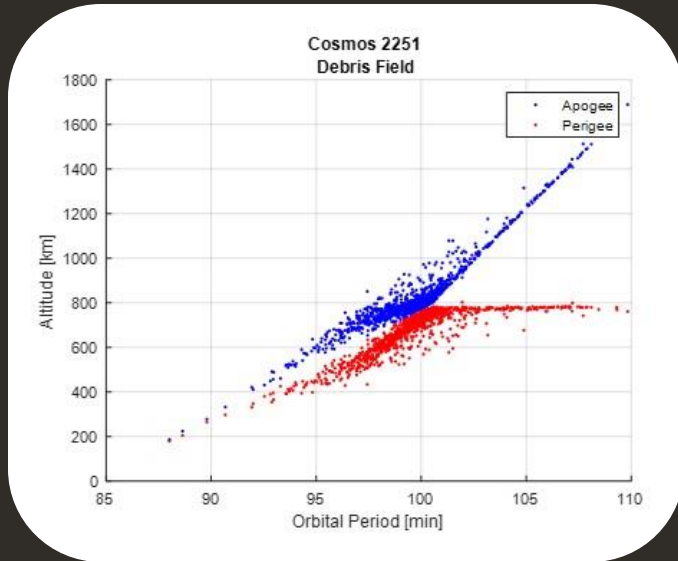


Centrifugal Pump Simulation

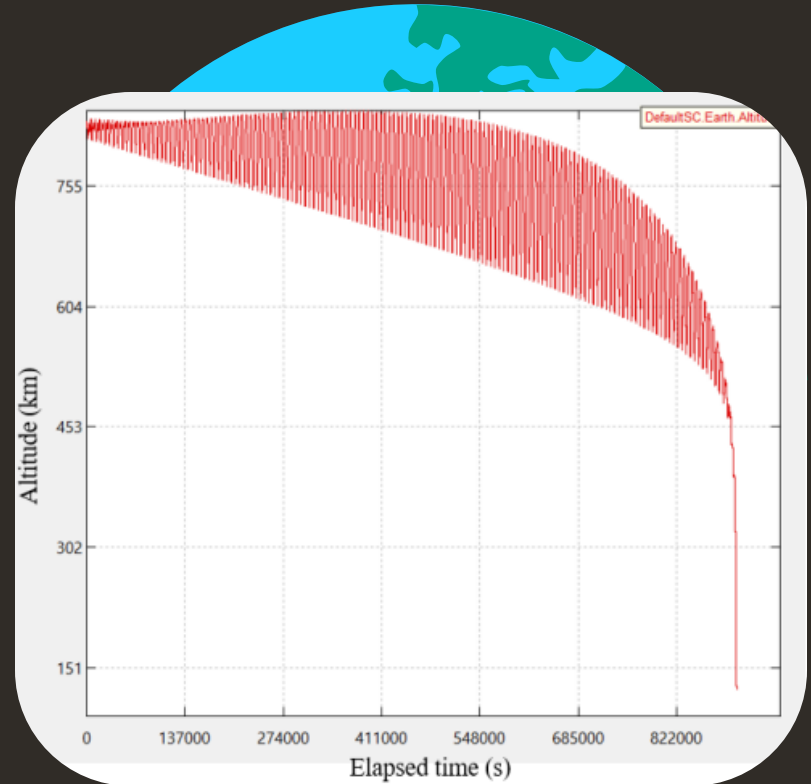


# Debris Mitigation

- Collision Consequence
- Kessler Syndrome
- Drag Augmentation reduces decay to <10 days



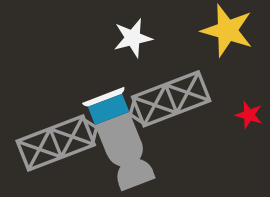
Debris spread from collision



Orbital Decay of CubeSat with EDT



# Structural Integrity and Material Selection

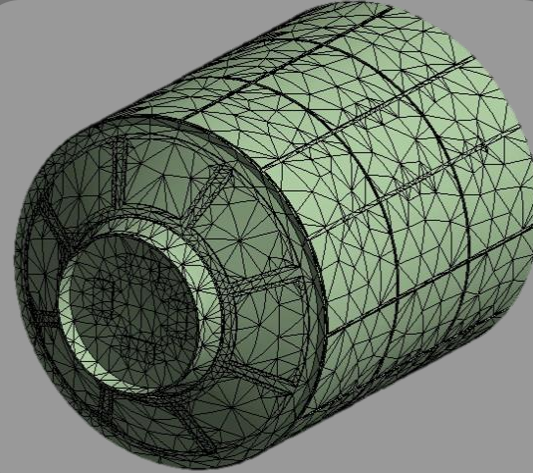


## Challenges

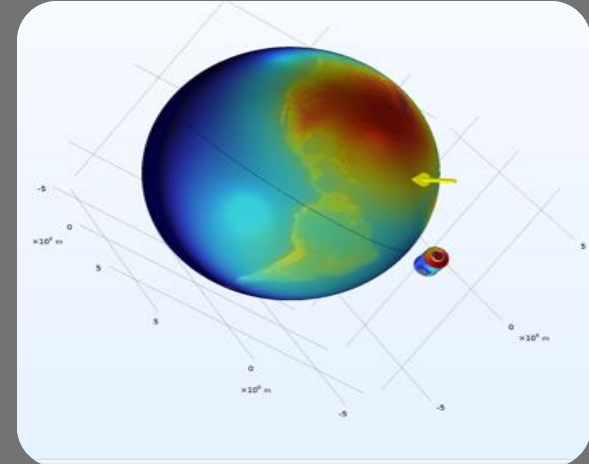
- Structure of our module
- Extreme Thermal conditions
- Micrometeorite impacts

## Experiments

- Whipple Shield Impact Simulation (Ansys)
- Thermal Variation Simulation (COMSOL Multi Physics)



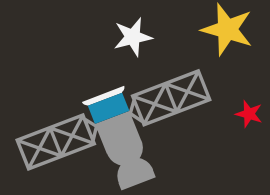
Module CAD Model



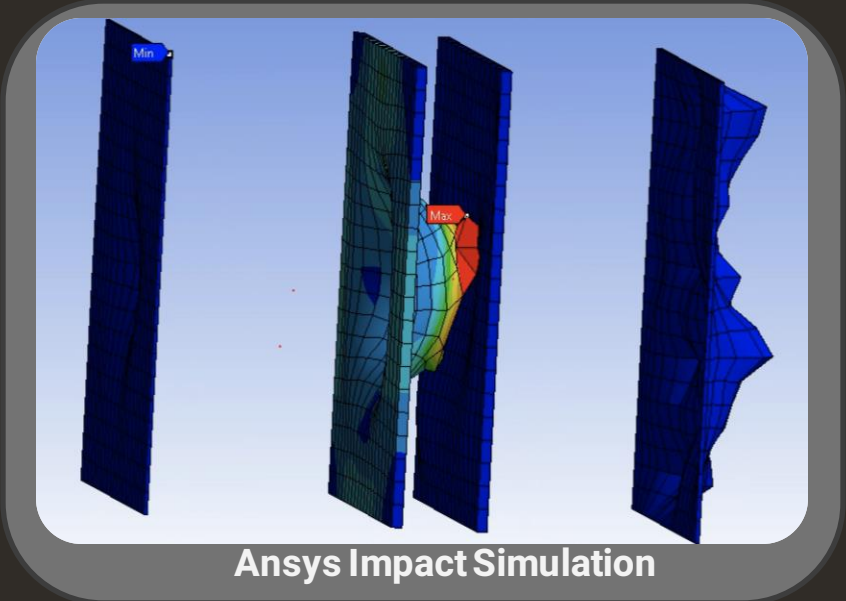
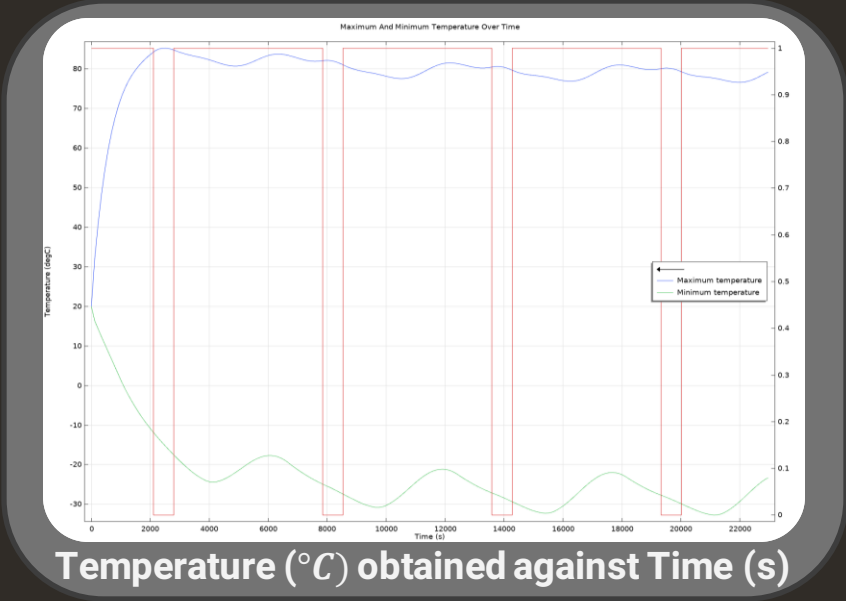
Thermal Variation Simulation



# Simulation

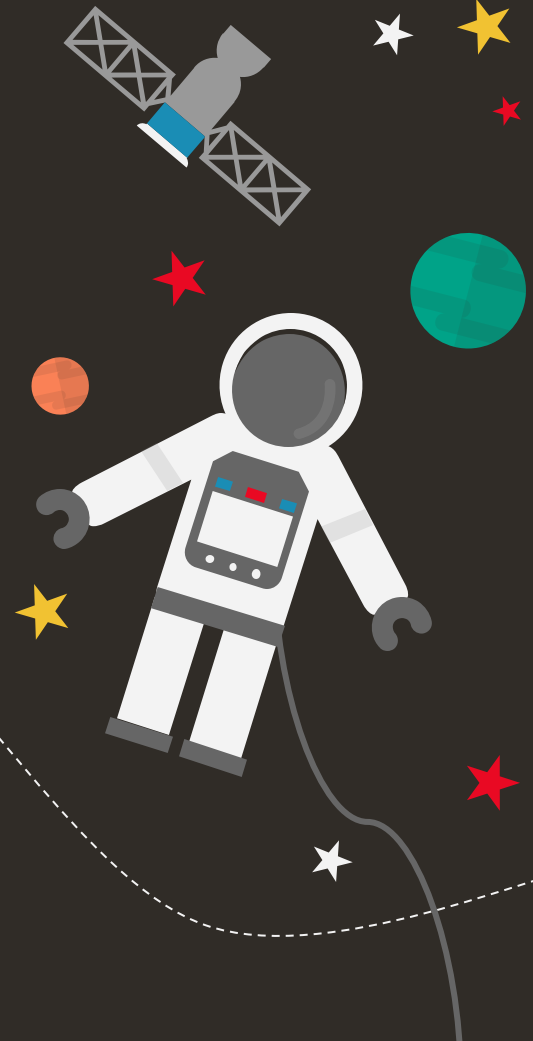


Chosen Material:  
Al 6061 + Spectra1000 + PI + Al 6061

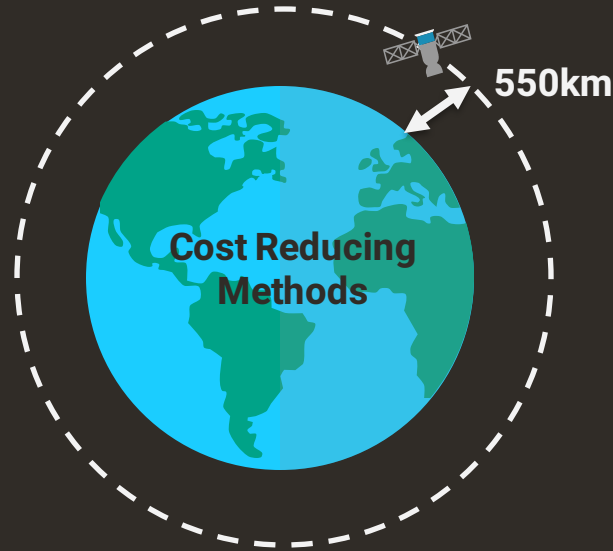


04

# Project Overview and Next Steps

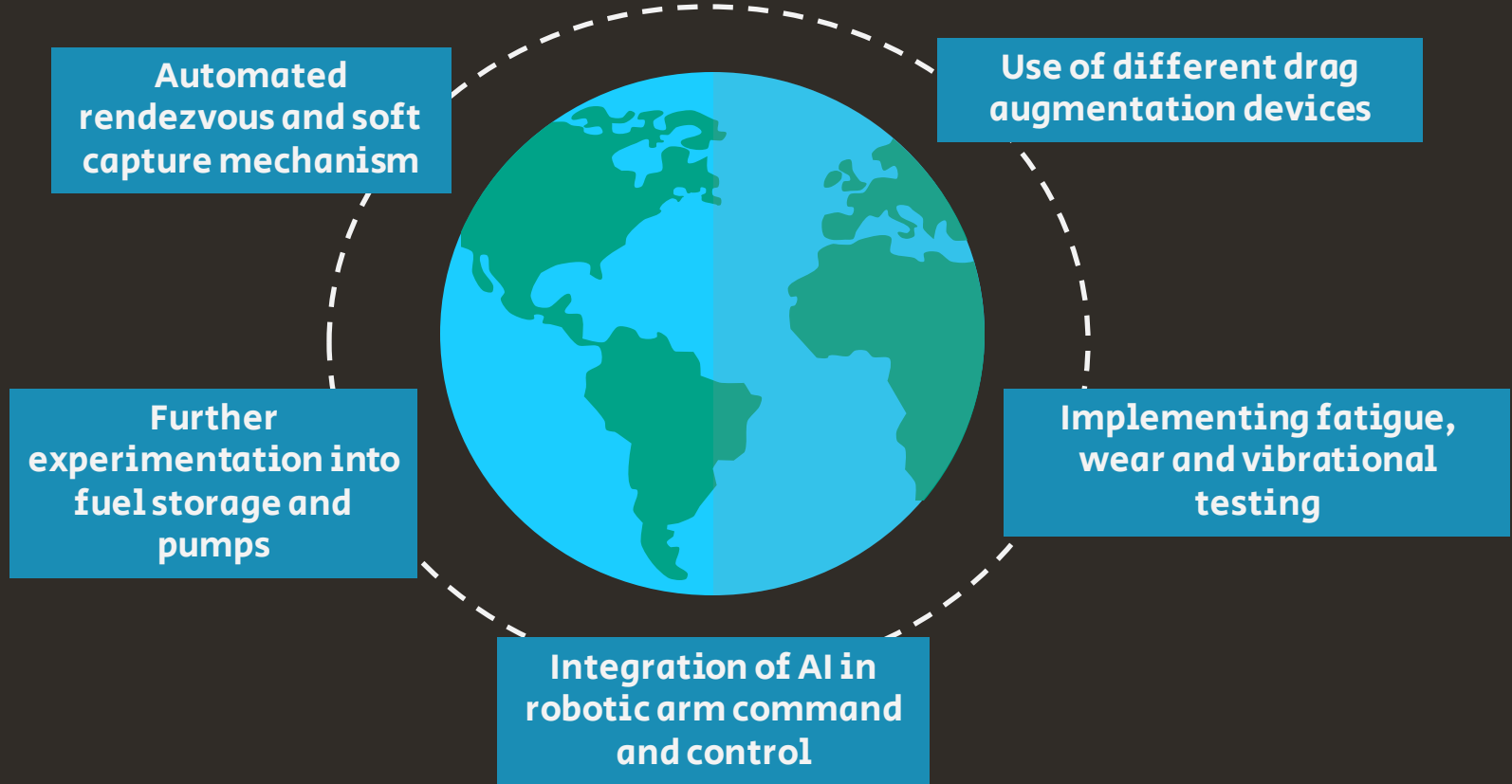


# What is Beyond Fuel?





# Further Improvements





# Thanks

Do you have any questions?

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